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NOISE MONITORING TITAN III D LAUNCH VANDENBERG AIR FORCE  
BASE, CALIFORNIA

Ronald D. Burnett

Environmental Health Laboratory  
McClellan Air Force Base, California

January 1975

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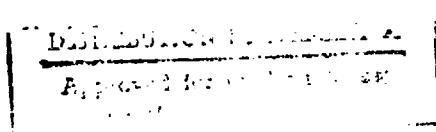
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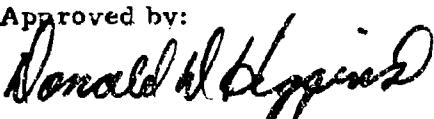
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## ABSTRACT

During a Titan III D launch at Vandenberg AFB, sound pressure levels were recorded on tape as a function of time at four distances varying from 8,400 ft to 44,000 ft from the launch site. One-third octave band data were obtained from the recordings using real time analysis techniques. Some limited vibration (acceleration) data were also collected at the La Purisima Mission, a historical park located approximately eleven miles from the launch site.

These data are discussed in relation to the environmental impact on communities surrounding Vandenberg AFB, and the damage potential of these launches to the La Purisima Mission State Historic Park. The environmental impact of noise at distances of eight miles or greater was insignificant.

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SECTION I  
INTRODUCTION

1. Purpose:

a. On 29 October 1974, noise data were collected at selected sites on and near Vandenberg AFB, California, during a Titan III D launch. This survey was requested by the bioenvironmental engineer, USAF Hospital, Vandenberg AFB, California. These data were required to estimate the acoustical impact of missile launches on communities adjacent to Vandenberg AFB. Limited vibration data were collected in the chapel of the La Purisima Mission State Historic Park to determine the extent of vibration induced in these historical structures during launches.

b. Because of the similarity between the Titan III D and launch vehicle systems for the space shuttle, these data are also to be used by personnel of the Space and Missile Systems Organization (SAMSO) to estimate the acoustical impact expected from space shuttle launches.

2. Scope of Study:

a. Overall and one-third octave band sound pressure levels as a function of time were measured at four sites. The frequency distribution of peak and background noise was measured at each site.

b. One-third octave band acceleration levels were measured on a roof beam of the La Purisima Mission Chapel, having the longest free span.

3. Personnel Contacted:

a. Lt Col Lynn R. Channell: Chief, Environmental Health Service, USAF Hospital, Vandenberg AFB.

b. Mr Clark Pease: Programs Support Manager, Programs Division, Range Operations, SAMTEC, Vandenberg.

c. Mr Mason: Area Director of State Historic Parks, La Purisima Mission.

**4. Personnel Conducting Survey & Survey Responsibilities:**

a. Maj Ronald B. Burnett, USAF Environmental Health Laboratory (USAFEHL), McClellan AFB: Project Engineer and Vibration Measurements.

b. Capt Larry Shingler, SAMSO/SGX, Los Angeles AFS: Noise Measurement at Oak Mountain Site.

c. 1st Lt Harry P. Guy, USAFEHL, McClellan AFB: Noise Measurement at Tranquillon Peak.

d. SSgt Ed Cox, USAFEHL, McClellan AFB: Noise Measurement at Range Operations (Building 488).

e. Mr Philip Diamond, USAFEHL, McClellan AFB: Noise Measurement at SLC-3 Blockhouse.

**SECTION II**

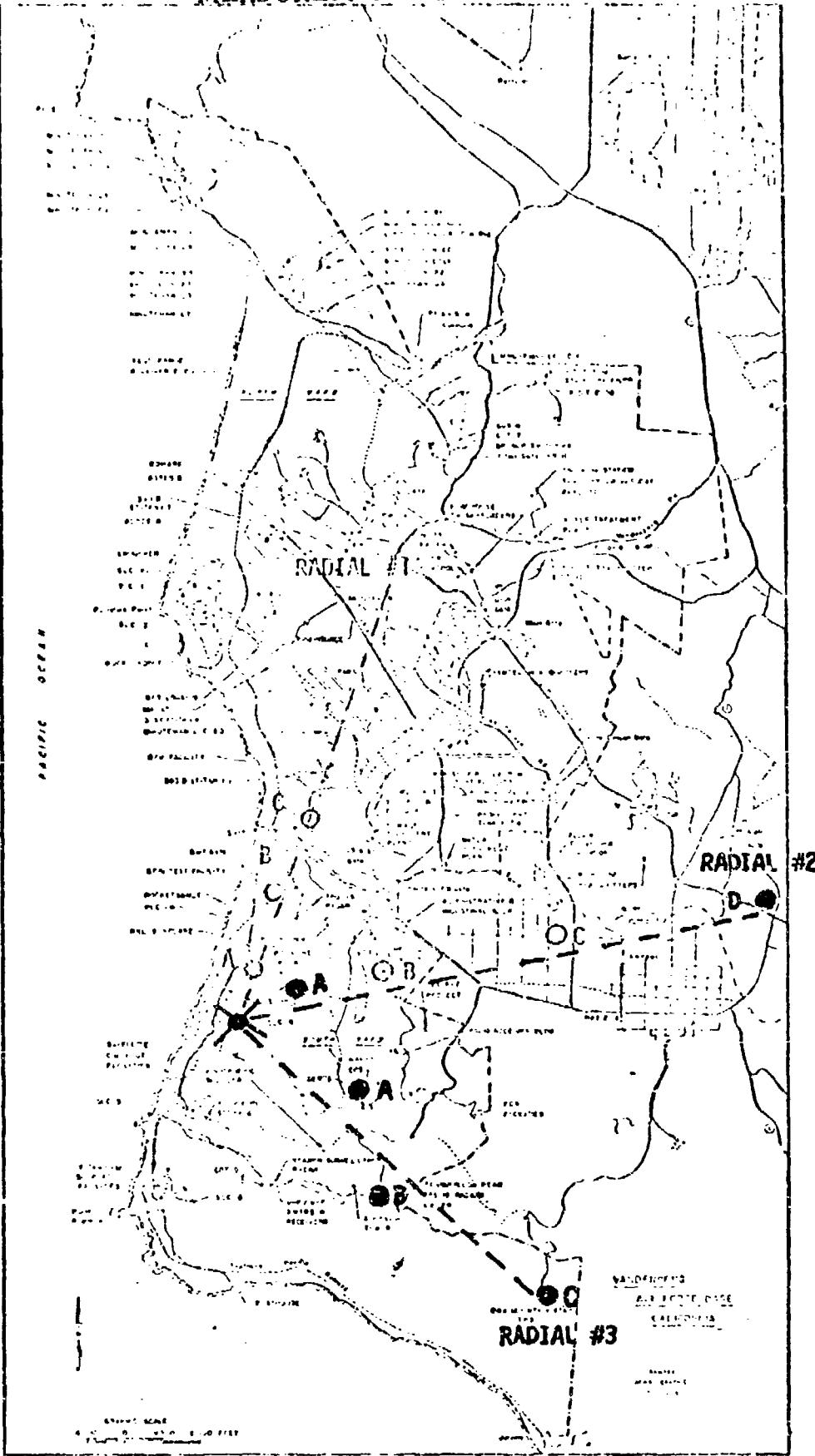
**SURVEY DESCRIPTION**

1. Measurement Sites. The location of the measurement sites in relation to the launch site are shown in Figure 1. Table I indicates the name of the measurement location and its designation on Figure 1.

**TABLE I**  
**IDENTIFICATION OF MEASUREMENT SITES**

<u>Measurement Site</u>	<u>Figure 1 Designation</u>	
	<u>Radial</u>	<u>Location</u>
SLC-3 Blockhouse	2	A
Range Operations (Bldg 488)	3	A
Tranquillon Peak	3	B
Oak Mountain	3	C
La Purisima Mission	2	D

FIGURE 1  
MEASUREMENT SITE LOCATIONS



## 2. Basic Measurement and Analysis Techniques:

### a. Noise.

(1) Measurement: Missile noise levels were recorded on Ampex 434 low noise tape and analyzed in the laboratory to obtain one-third octave band sound pressure levels as a function of time. A 114 decibel (dB), 1000 Hertz (Hz) calibration tone was recorded prior to and following the actual noise measurements. The frequency distribution (1/3 octave bands) of peak noise levels was also determined. Overall sound pressure levels were calculated from the 1/3 octave band data and presented as a function of time. A block diagram showing the measurement equipment set up is presented in Figure 2.

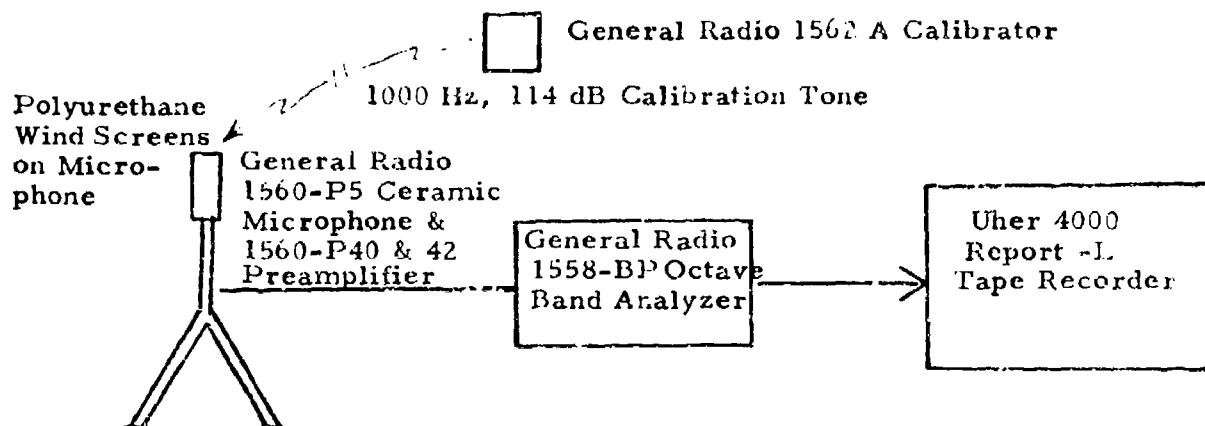
(2) Analysis: The taped data were analyzed using a real time noise and vibration analysis system. The 114 dB, 1000 Hz, calibration tone recorded on each tape was used as a reference level for analysis of the taped noise. A frequency response curve for each recording and playback system was also developed to align data output during analysis in the laboratory. The recording and analysis systems accuracy was determined by a simplified uncertainty analysis (see Appendix A) and the uncertainty (limit of error) in the resulting data was  $\pm$  1.9 dB. A block diagram of the noise analysis system is shown in Figure 3.

### b. Vibration:

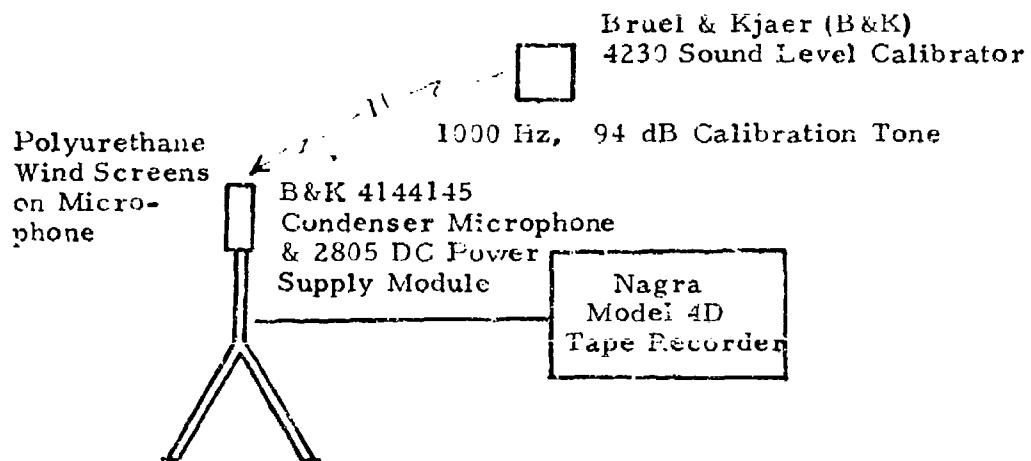
(1) Measurement: Two vibration pickups (accelerometers) were placed on a roof beam appearing to have the longest unsupported length. Vibration data (acceleration) were recorded on a dual channel tape recorder. The taped data were analyzed in the laboratory to obtain acceleration and frequency distribution data. Figure 4 shows a block diagram of the vibration measurement system.

(2) Analysis: The taped data were analyzed using the real time noise and vibration analysis system. A one g ( $g$  = acceleration of gravity), 100 Hz, calibration signal was recorded on the tape to be used as a reference signal during analysis of the taped vibration data. A frequency response curve for each measurement and playback channel was also developed to align the data output during analysis. The vibration analysis system was identical to that used for noise analysis (Figure 3).

FIGURE 2  
NOISE MEASUREMENT SYSTEMS



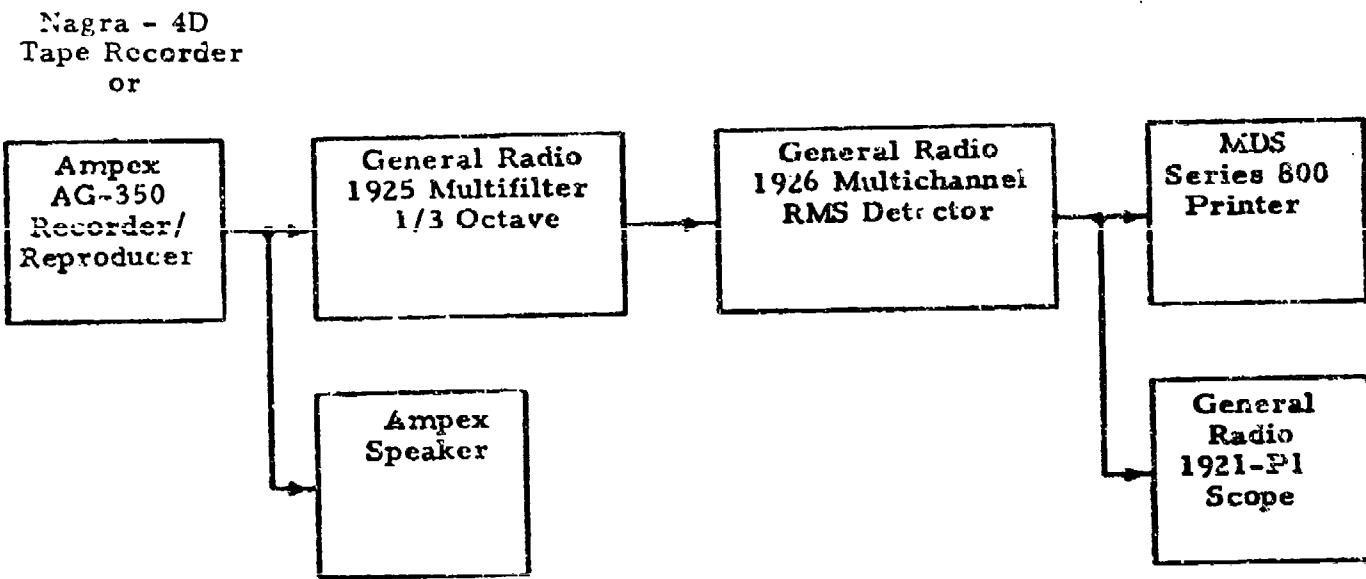
This system used for all sites except Tranquillon Peak



This system was used at Tranquillon Peak

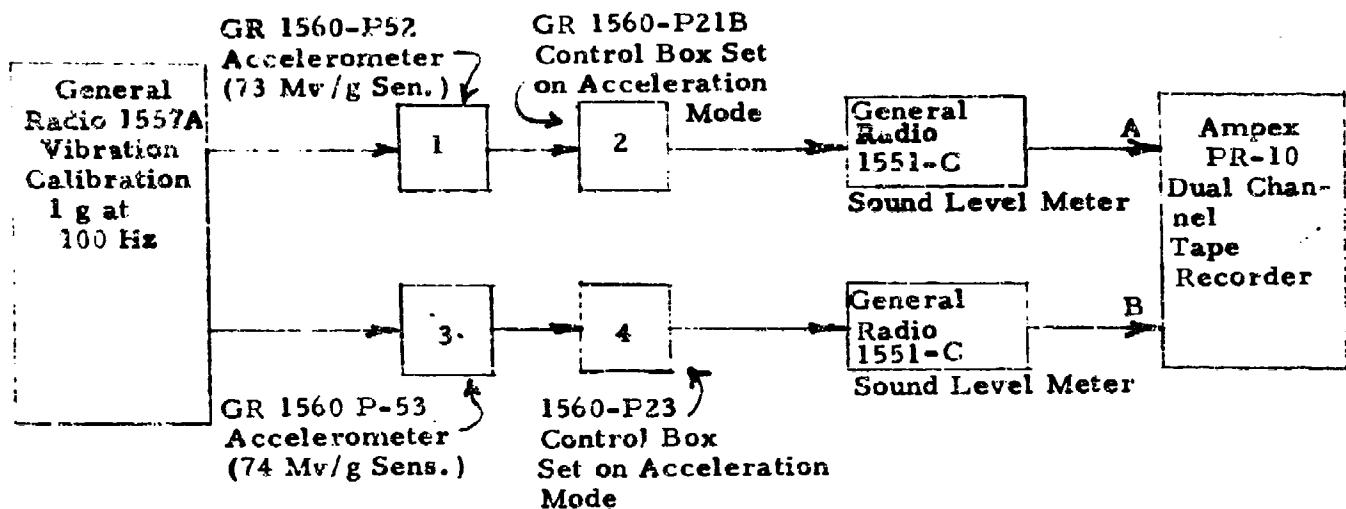
FIGURE 3  
NOISE ANALYSIS SYSTEM

Note: Only the noise data taped at Tranquillon Peak were played back for analysis with the Nagra recorder. The other tapes were played back with the Ampex AG-350.



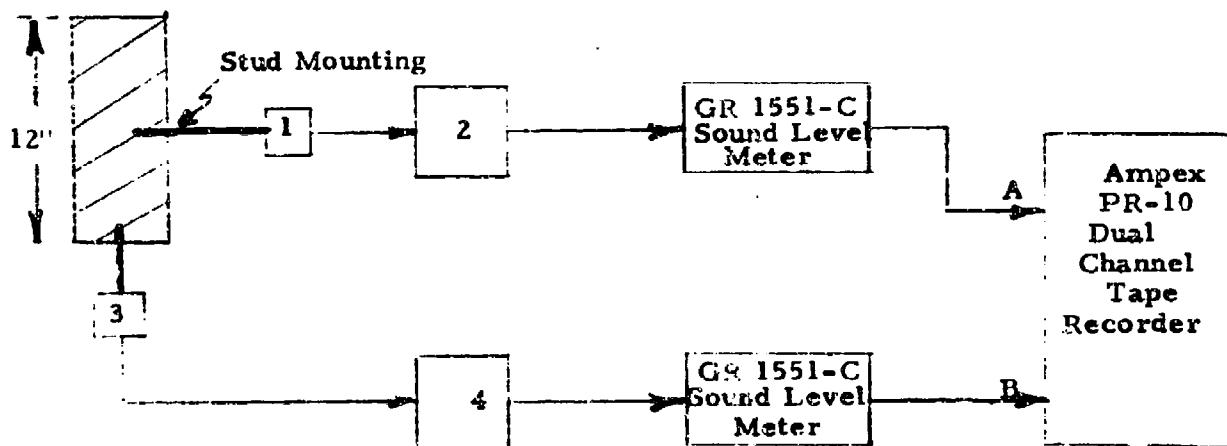
**FIGURE 4**  
**VIBRATION MEASUREMENTS**

Calibration



Measurement

2" x 12" Beam Cross Section



A = Channel A

7

B = Channel B

## SECTION III

### DISCUSSION AND RESULTS

1. General: The predominantly low frequency noise, characteristic of rocket engines, was difficult to measure and analyze with the laboratory's noise analysis systems which are normally used for a frequency range of 25 Hz to 10 KHz. However, by recording the noise levels at 7 1/2 inches/second (ips) and playing them back at 15 ips, which essentially doubled the frequency of the recorded noise, it was possible to obtain 1/3 octave band sound pressure level data from 15.75 Hz to 8 KHz. The vibration data were limited by the frequency characteristics of the measuring equipment to a range of 25 Hz to 2 KHz. These equipment limitations, the lack of experience of laboratory personnel in measuring and evaluating vibration, coupled with a lack of existing vibration criteria specifying some level at which structural damage might be expected to occur, severely limits the confidence which can be placed upon the vibration data and their interpretation.

2. Noise:

a. One-third octave band and overall sound pressure levels: These data are presented as a function of time in Tables II thru V. Since the exact noise levels occurring at each site were not precisely known prior to the launch it was necessary to adjust the sound level measuring and recording equipment such that expected peak levels would not overdrive the instrumentation and saturate the tapes. This procedure increased the noise floor of the measurement instrumentation. Therefore, the lower noise levels measured in the high frequencies and during the first and last seconds of the launch may represent inherent instrumentation noise rather than actual ambient noise levels. Where this was clearly the case the levels in the Tables are identified by less than or equal signs ( $\leq$ ). The values presented in the Tables are root mean squared (RMS) sound pressure levels over a sample time of 7.6 seconds (8 seconds at Tranquillon Peak site). Although the real time analysis system is capable of integrating noise data in each frequency band every 1/8 seconds, the data printer associated with the system cannot assimilate the data this rapidly; therefore, a 4.0 second integration time was used. This is equivalent to 3.8 seconds actual launch time because there is a slight difference between the record speed of the Uher field recorder and the playback speed of the Ampex laboratory reproduction system. Thus, at 15 ips playback speed, 7.6 seconds of actual range time elapsed during the 4 second data integration period. The Tranquillon Peak data were played back with the

TABLE II  
SLC-3 BLOCKHOUSE  
SOUND PRESSURE LEVELS AND FREQUENCY DIS-  
AS A FUNCTION OF TIME

Time (Sec)	1/3 OCTAVE BAND (Hz) LEVELS (dB re 20																		
	15.75	20	25	31.5	40	50	63	80	100	125	160	200	250	315	400	500	630	800	1.0k
0	86	83	78	76	76	74	72	68	67	68	69	77	65	71	≤65	≤65	≤65	≤65	≤65
7.7	102	105	107	107	106	103	105	101	96	93	93	92	87	88	89	90	88	89	88
15.4	112	113	116	116	116	114	113	110	108	106	104	101	100	101	102	101	98	98	96
23.1	115	112	114	113	112	112	112	111	106	101	101	103	103	103	99	97	96	95	94
30.1	105	105	109	110	109	107	106	103	96	98	100	101	98	94	94	93	90	90	89
38.5	99	103	106	104	99	97	93	88	88	93	93	93	85	88	86	85	83	83	82
46.2	97	97	99	98	96	90	85	82	85	90	90	88	80	86	80	81	77	74	71
53.9	92	91	92	92	90	82	82	79	77	81	86	85	83	76	80	73	74	70	67
61.5	90	88	86	85	84	76	73	71	73	76	74	77	68	73	≤65	≤65	≤65	≤65	≤65
69.2	87	82	83	81	80	75	76	72	71	71	71	78	65	72					
76.9	84	78	80	76	76	73	73	71	71	73	71	77		72					
84.6	82	76	74	75	72	72	69	68	68	69	69	77		71					
92.3	80	79	77	76	74	74	73	71	70	70	70	77	71	71	71	71	71	71	71
Background	67	64	61	57	55	51	55	58	58	64	53	47	53	44	38	42	41	42	42

TABLE II

## **SLC-3 BLOCKHOUSE**

## **SURE LEVELS AND FREQUENCY DISTRIBUTION AS A FUNCTION OF TIME**

### CTAVE BAND (Hz) LEVELS (dB re 20 $\mu$ N/m<sup>2</sup>)

200	250	315	400	500	630	800	1.0k	1.25k	1.6k	2.0k	2.5k	3.15k	4.0k	5.0k	6.3k	8.0k	Overall SPL
77	65	71	≤65	≤65	≤65	≤65	≤65	≤65	≤65	≤65	≤65	≤65	≤65	≤65	≤65	≤65	90
92	87	88	89	90	88	89	88	87	79	76	72	68	68	68	70	70	114
101	100	101	102	101	98	98	96	95	92	92	90	89	89	89	84	81	124
103	103	103	99	97	96	95	94	92	91	89	88	86	84	81	79	78	122
101	98	94	94	93	90	90	89	88	86	84	83	81	80	76	73	80	117
93	85	88	86	85	83	83	82	79	78	76	74	71	69	69	70	70	111
88	80	86	80	81	77	74	71	68	68	67	65	65	66	67	70	70	105
85	83	76	80	73	74	70	67	≤65	≤65	≤65	≤65	≤65	≤65	≤65	≤65	≤65	99
77	68	73	≤65	≤65	≤65	≤65	≤65							≤66	≤67	70	95
78	65	72															91
77		72															88
77		71															86
77	↑	71	↓	↓	↑	↑	↑	↑	↑	↑	↑	↑	↑	↑	↑	↑	87
47	53	44	38	42	41	42	42	42	39	39	37	34	34	30	29	28	72

TABLE III  
RANGE OPERATIONS (BLD)  
SOUND PRESSURE LEVELS AND FREQUENCY DE  
AS A FUNCTION OF TIME

1/3 OCTAVE BAND (Hz) LEVELS (dB re 20μ)

	15	75	20	25	31.5	40	50	63	80	100	125	160	200	250	315	400	500	630	800	1,000
0	76	73	70	70	63	63	65	58	59	560	560	558	560	564	560	559	559	560	560	561
7.6	80	78	74	60	64	69	72	61	62	61	63	62	61	564	60	60	559	560	560	561
15.2	91	93	86	78	88	94	99	92	93	91	89	86	84	85	84	80	78	78	78	77
22.9	107	106	103	90	105	110	111	99	101	101	99	94	94	94	92	90	88	88	87	
30.5	113	112	105	93	107	112	114	106	102	106	103	98	97	93	94	93	92	91	91	89
37.0	115	112	106	95	104	109	111	103	101	103	96	94	97	99	98	93	90	90	88	
45.7	108	106	102	93	99	103	107	99	92	92	88	94	97	94	86	87	87	85	85	
53.3	106	102	100	92	95	102	106	97	88	85	92	93	94	89	88	87	83	82	80	
60.9	101	98	98	89	90	97	100	91	86	85	94	92	92	85	87	82	80	75	73	
68.6	99	96	94	86	87	92	94	86	80	80	87	85	81	76	76	70	67	64	562	
76.2	97	94	91	84	85	88	88	80	73	78	80	80	74	72	72	65	63	560	561	
83.8	94	91	87	79	84	83	85	77	71	74	75	73	68	66	64	60	559	559	560	
91.4	91	89	85	80	83	83	84	76	70	75	76	74	71	65	63	559	560	560	560	
99.1	89	87	85	78	82	82	83	77		72	74	72	69	65	61		560	560	560	
106.7	89	85	83	77	81	80	82	75		72	75	74	72	564	560		560	560	560	
114.3	85	83	84	76	80	78	81	72		70	72	69	67	564	560		560	561	561	
121.9	85	84	85	76	80	79	82	72		72	74	72	69	564	560		559	560	560	
129.5	83	85	82	75	78	77	79	70		70	71	68	66	564	561		560	561	561	
138.5	81	84	81	73	77	75	76	67	64	65	68	66	63	563	560			561	561	
144.8	83	79	80	72	75	74	75	65	64	63	65	64	62	563	560	558		561	561	
152.4	79	79	79	70	73	74	74	64	64	61	64	62	62	563	560	559		560	560	
160.0	73	77	76	70	72	71	72	61	61	61	62	62	561	564	559	560		561	561	
167.6	76	76	76	70	72	72	72	63	62	61	63	62	561	563	560	559		560	560	
175.2	78	76	76	70	72	74	72	63	63	560	62	562	561	564	559	558				
182.9	76	76	78	72	72	73	71	63	62	560	561	559	560	563	559	558				
190.5	75	75	76	71	71	74	72	63	64	562	561	561	561	564	560	559				
198.1	74	71	73	64	69	73	70	60	64	563	560	560	561	564	560	558	558			
205.7	76	75	77	68	78	71	70	62	61	560	560	561	560	563	559	559	559	559		
213.3	71	71	72	66	67	68	70	61	60	560	560	558	560	563	559	558	559	559	559	
Back Ground	78	74	74	63	64	63	62	53	51	51	46	44	44	42	41	40	39	40	41	

TABLE III

## RANGE OPERATIONS (BLDG 488)

SIRE LEVELS AND FREQUENCY DISTRIBUTION  
AS A FUNCTION OF TIMEAVERAGE BAND (Hz) LEVELS (dB re 20 $\mu$ N/m<sup>2</sup>)

0	250	315	400	500	630	800	1.0k	1.25k	1.6k	2.0k	2.5k	3.15k	4.0k	5.0k	6.3k	8.0k	Overall SIRE
8	≤60	≤64	≤60	≤59	≤59	≤60	≤61	≤61	≤62	≤63	≤65	≤64	≤65	≤66	≤64	≤63	81
2	61	≤64	60	60	≤59	≤60	≤61	≤62	≤62	≤63	≤65	≤65	≤65	≤65	≤65		84
6	84	85	84	80	78	78	77	73	71	67	≤65	≤65	≤64	≤65	≤65		103
4	94	94	92	90	88	88	87	85	82	80	78	74	68	67	≤65		116
8	97	93	94	93	92	91	89	88	86	84	83	78	74	71	68	≤65	120
4	97	99	98	93	90	90	88	86	87	85	84	80	76	71	68	≤64	119
4	97	94	86	87	87	85	85	84	81	80	78	74	71	68	≤65	≤63	113
3	94	89	88	87	83	82	80	78	77	74	72	66	≤66	≤66	≤64		111
2	92	85	87	82	80	75	73	71	71	71	70	67	≤65	≤65	≤64		107
5	81	76	76	70	67	64	≤62	≤62	≤62	≤63	≤65	≤64			≤64		103
0	74	72	72	65	63	≤60	≤61	≤61	≤62			≤65			≤65		100
3	68	66	64	60	≤59	≤59	≤60	≤61	≤62			≤64			≤64		97
4	71	65	63	≤59		≤60	≤60	≤61	≤61						≤64		95
2	69	65	61		≤60	≤60	≤61	≤61							≤64	≤62	94
4	72	≤64	≤60		≤60	≤60	≤61	≤62		≤64					≤64	≤63	93
9	67	≤64	≤60		≤60	≤61	≤62	≤62		≤65		≤64		≤63	≤62		91
2	69	≤64	≤60		≤59	≤60	≤61	≤61	≤62			≤65		≤64	≤62		92
8	66	≤64	≤61		≤60	≤61	≤61	≤62	≤64			≤64	≤64	≤63	≤61		90
6	63	≤63	≤60		↓	≤61	≤61	≤62	≤63			≤64	≤65	≤63	≤62		88
4	62	≤63	≤60	≤58		≤61	≤61	≤61				≤65	≤64		≤61		87
2	62	≤63	≤60	≤59		≤60	≤62	≤62				≤64	≤65		≤62		86
2	≤61	≤64	≤59	≤60		≤61	≤61	≤62		↓	↓		≤64		≤61		83
2	≤61	≤63	≤60	≤59		≤60		≤62		≤64	≤65				≤65		83
2	≤61	≤64	≤59	≤58			≤61		≤65	≤64	≤65	≤65			↓		84
9	≤60	≤63	≤59	≤58			≤61		≤64		≤64	≤64			≤62		84
1	≤61	≤64	≤60	≤59	↓			≤62							≤61		83
0	≤61	≤64	≤60	≤58	≤58			≤62							≤62		81
1	≤60	≤63	≤59	≤59	≤59	↓		≤62							↓	≤61	83
8	≤60	≤63	≤59	≤58	≤59	≤59	↓	↓	≤61	↓	↓	↓	↓	↓	≤62	≤61	80
4	44	42	41	40	39	40	41	42	42	44	45	45	45	46	44	42	81

TABLE IV

## TRANQUILLON PEAK

SOUND PRESSURE LEVELS AND FREQUENCY DE  
AS A FUNCTION OF TIME1/3 OCTAVE BAND (Hz) LEVELS (dB re 20  $\mu$ Pa)

	10.75	20	25	31.5	40	50	63	80	100	125	160	200	250	315	400	500	630	800	1.0k	1.
0	69	67	65	60	58	57	57	56	55	53	50	63	55	57	52	50	50	51	52	
8	68	66	60	60	56	55	54	52	53	52	50	62	54	57	53	51	50	51	52	
16	74	74	74	74	72	73	70	70	69	68	69	65	61	61	60	60	57	54	53	
24	88	90	90	93	94	94	92	92	91	90	89	85	83	81	78	77	75	71	68	
32	99	100	99	102	102	100	99	100	100	96	96	91	87	87	84	83	81	76	73	
40	95	95	95	96	99	96	94	95	94	90	88	85	84	84	83	81	78	78	75	
48	100	100	98	100	102	99	99	99	96	91	89	89	90	91	86	83	85	82	82	
56	104	103	101	101	102	99	98	95	95	89	92	94	95	90	87	86	84	84	83	
64	99	100	98	99	98	94	93	92	88	84	90	89	88	81	84	80	78	77	75	
72	97	97	96	96	93	92	91	87	82	82	87	85	82	78	81	75	74	72	70	
80	93	94	94	91	91	91	87	83	80	81	85	81	78	76	75	71	68	62	59	
88	92	93	90	88	89	88	83	79	77	78	81	76	73	72	72	67	62	57	56	
96	91	92	88	88	87	86	83	78	76	78	80	75	70	69	69	61	56	53	52	
104	89	89	84	86	83	80	76	73	73	76	71	68	66	65	64	59	53	51	51	
112	87	86	84	85	83	81	77	74	72	72	76	71	69	64	65	58	53			
120	87	82	82	82	81	78	74	73	69	68	70	67	60	59	56	52	50			
128	85	83	82	81	80	79	74	71	69	68	72	68	64	60	58	50				
136	87	84	82	82	80	77	73	72	67	67	71	84	62	59	57	51	50			
144	85	82	79	81	78	74	71	70	67	63	67	66	60	58	54	51	50	51		
152	83	80	77	78	76	73	72	69	66	62	66	65	60	58	55	51	50	50		
160	82	78	75	77	76	73	72	67	66	60	62	63	57	57	53	50	49	51		
168	82	78	75	76	75	70	67	64	62	60	60					53	51	50		
176	78	78	73	75	73	71	68	67	62	60	62					54	50	50		
184	79	77	75	76	75	71	68	66	60	57	58	62				53	50	49	50	
192	80	75	73	75	72	70	68	66	61	59	60	63	58	58		51	50	50		
200	77	74	70	73	70	64	52	62	61	59	58	63	56	57		51	50	51		
208	77	74	70	73	72	68	66	61	56	54	52	52	55			50	50	51		
216	77	74	70	73	70	69	66	59	58	55	54	51	55			50	50	50		
Back Ground	70	71	63	62	61	60	60	53	51	51	49	49	49	49	48	47	47	47	48	

TABLE IV

## TRANQUILLON PEAK

PRESSURE LEVELS AND FREQUENCY DISTRIBUTION  
AS A FUNCTION OF TIMEAVERAGE BAND (Hz) LEVELS (dB re 20  $\mu\text{N}/\text{m}^2$ )

200	250	315	400	500	630	800	1.0k	1.25k	1.6k	2.0k	2.5k	3.15k	4.0k	5.0k	6.3k	8.0k	Overall SPL
63	55	57	52	50	50	51	52	52	55	55	54	56	56	57	57	60	74
62	54	57	53	51	50	51	52	52	55	55		57		57			73
65	61	61	60	60	57	54	53	53	55	54		56		58			83
85	83	81	78	77	75	71	68	62	60	55		57		61			102
91	87	87	84	83	81	76	73	68	65	59		57	58	61			110
85	84	84	83	81	78	78	75	72	68	66	60	58	57	60			106
89	90	91	86	83	85	82	82	79	77	75	71	67	65	61			109
94	95	90	87	86	84	84	83	79	77	79	74	68	71	62	59		111
89	88	81	84	80	78	77	75	73	68	66	61	60	58	58			107
85	82	78	81	75	74	72	70	68	65	60	55	57	57	57			104
81	78	76	75	71	68	62	59	55	57	55	54	55	56		60		101
76	73	72	72	67	62	57	56	53	55	55		56	57		61		99
75	70	69	69	61	56	53	52	53		54				57	60		97
68	66	65	64	59	53	51		52						58	61		94
71	69	64	65	58	53								58	60			93
67	60	59	56	52	50				55			56		57			91
68	64	60	58	50	50				53	54				58			90
84	62	59	57	51	50	50		53						57			92
66	60	58	54	51	50	51		52			54			58			89
65	60	58	55	51	50	50				55	55	57		57			87
63	57	57	53	50	49	51				54	56			58			86
			53	51	50				55	54				57			85
			54	50	50				54	53				58			84
562	1	1	53	50	49	50		53			54	55	56	57			84
63	58	58	51	50	50	50					56	56	58				84
63	56	57	51	50	51				57			57		57			81
62	55	1	50	50	51			52	54	55		56	58	58			82
61	55	1	50	50	50			52	55	54	Y	56	57	57	58		81
49	49	49	48	47	47	47	48	48	48	47	47	47	47	48	47	46	75

**TABLE V**  
**OAK MOUNTAIN**  
**SOUND PRESSURE LEVELS AND FREQUENCY D**  
**AS A FUNCTION OF TIME**

		1/3 OCTAVE BAND (Hz) LEVELS (dB re 20 µPa)																			
Frequency (Hz)		13.75	20	25	31.5	40	50	63	80	100	125	160	200	250	315	400	500	630	800	1.0k	1.5k
0	70	65	61	62	59	63	59	56	55	58	54	51	50	57	48	47	47	47	47	47	47
7.7	72	67	63	63	61	63	60	59	59	60	57	54	51	57	49		48				
15.4	71	68	64	64	62	64	62	61	59	60	56	54	51	58	49		47				
23.1	79	78	72	73	69	67	62	59	56	57	55	50	49	56	48	46					
30.8	82	78	74	74	70	69	63	61	58	60	56	53	50	58	49	48	47				
38.5	76	74	73	73	76	75	72	74	74	72	69	69	63	61	55	52	50	48	47		
46.2	85	80	80	83	86	85	84	86	85	82	80	77	73	69	63	61	60	56	51		
53.9	95	95	95	96	97	96	95	95	93	91	89	86	81	77	72	69	66	63	58		
61.5	93	91	91	95	95	97	96	96	95	93	91	88	83	79	75	73	71	68	63		
69.2	92	90	92	95	95	93	93	92	89	87	85	81	76	78	78	75	68	67			
76.9	100	100	98	101	100	98	96	94	93	90	87	81	81	85	86	84	80	74	74		
84.6	99	96	98	100	99	97	96	93	89	87	81	82	84	87	84	77	77	74	72		
92.3	97	96	95	98	97	95	92	89	86	81	80	82	83	83	77	74	74	68	66		
100.0	93	95	93	96	94	93	90	86	81	75	77	79	78	76	66	68	64	61	53		
107.7	94	92	90	93	92	91	88	84	77	74	75	78	77	72	64	64	56	54	49		
115.4	95	95	92	90	92	91	90	87	83	75	74	77	80	77	70	64	53	50	48		
123.1	93	91	88	89	89	87	85	80	73	72	76	78	74	66	66	59	53	48	47		
130.8	90	91	87	89	87	86	83	77	70	72	75	76	71	63	64	54	51	49	51		
138.5	89	88	86	84	84	83	80	75	70	71	73	74	69	61	60	53	49	47	47		
146.2	90	88	84	84	82	81	79	75	68	71	71	72	67	59	56	51	48	48			
153.9	87	85	83	82	80	79	77	73	66	69	70	71	65	60	58	51	48	47			
161.5	85	82	79	80	77	75	74	72	65	67	68	68	62	58	56	49	47				
169.2	86	83	81	81	79	76	75	72	67	66	69	69	64	58	53	48					
176.9	87	84	80	82	80	76	75	73	67	65	67	68	62	57	53	48	47				
Background	83	81	73	76	71	65	62	60	58	57	49	46	43	43	41	40	40	40	40	40	

TABLE V

## OAK MOUNTAIN

## SURE LEVELS AND FREQUENCY DISTRIBUTION AS A FUNCTION OF TIME

### AVE BAND (Hz) LEVELS (dB re 20 $\mu$ N/m<sup>2</sup>)

field recorder (see Figure 2), thus record and reproduce tape speeds were the same. At this site a 4 second integration time was equal to an actual 4 seconds of launch or range time at 7 1/2 ips and 8 seconds of range time at 15 ips playback speed.

b. Peak Noise and Frequency Distribution: The time that peak noise levels occurred was determined from the overall levels presented in Tables II thru V. The tapes were then analyzed at that time using a 1/8 second integration time to obtain peak instantaneous noise levels and their frequency distribution. Figures 5 thru 8 present peak noise levels as a function of frequency (1/3 octave bands). The peak levels and frequency distribution are a little more accurate because the averaging time, i.e., integration time (1/8 second) was much shorter.

c. Comparison of Measured With Predicted Noise Levels: The noise levels measured were within the range predicted on the basis of Titan III C, D and E noise data provided by SAMSO/LVRG personnel (see Appendix B). Table VI shows measured and predicted noise levels at each measurement site.

TABLE VI  
COMPARISON OF MEASURED & PREDICTED LEVELS

<u>Location</u>	<u>Ground Distance From Launch Site ft)</u>	<u>Peak OSPL'S (dB) re 20 <math>\mu</math> N/m<sup>2</sup></u>	<u>Predicted</u>	<u>Measured</u>
SLC-3	8,400		124	124
Range Ops (Bldg 488)	16,200		118	120
Tranquillon Peak	24,000		114	111
Oak Mountain	44,000		107	108

These overall peak sound pressure levels are plotted as a function of distance in Figure 9 to facilitate extrapolation of these data to distances greater than measured. Since low frequency noise is difficult to attenuate, the use of existing data for application at different

FIGURE 5

SLC-3 BLOCKHOUSE  
PEAK SOUND PRESSURE LEVELS

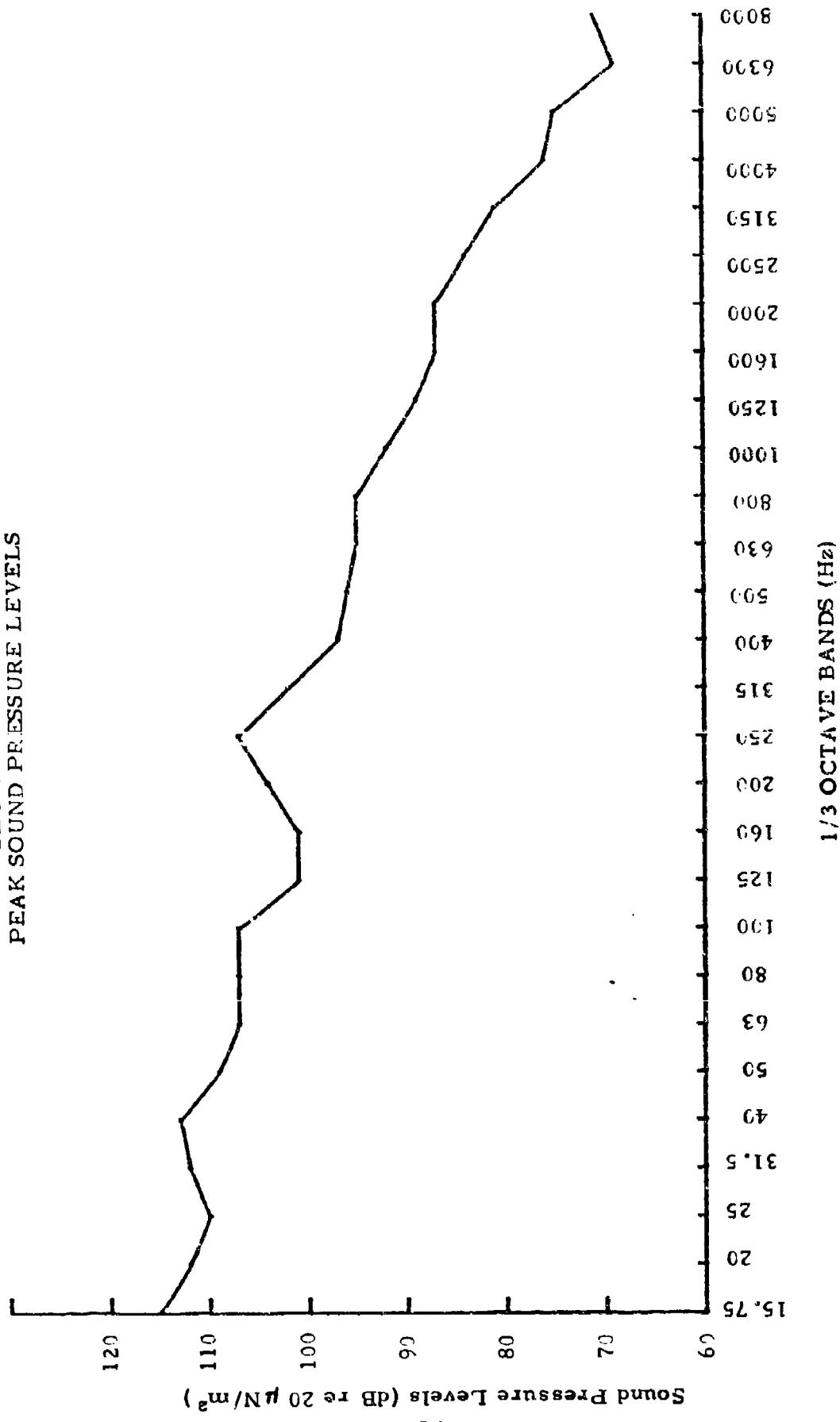


FIGURE 6

RANGE OPERATIONS  
PEAK SOUND PRESSURE LEVELS

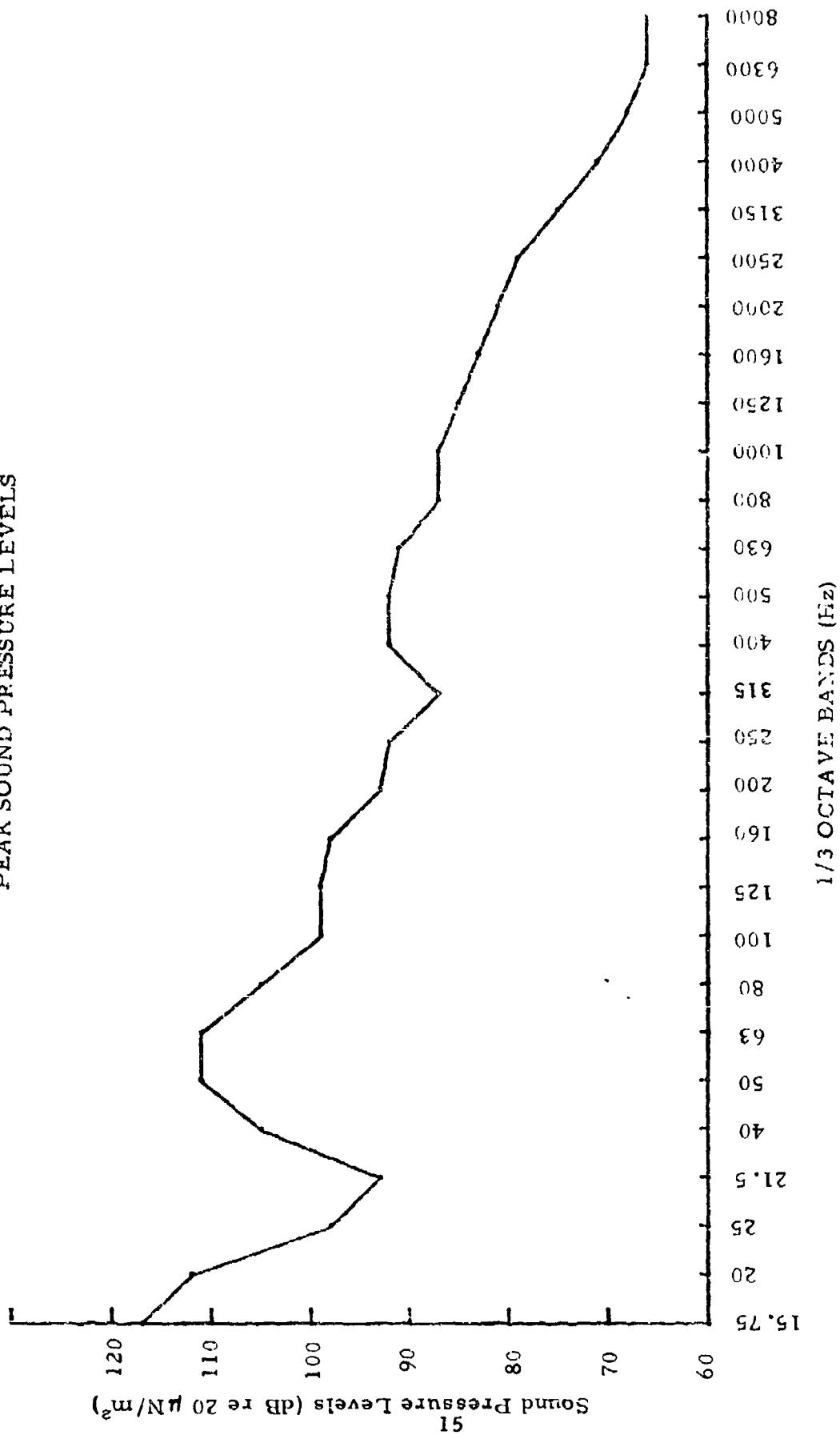


FIGURE 7

TRANQUILLON PEAK  
PEAK SOUND PRESSURE LEVELS

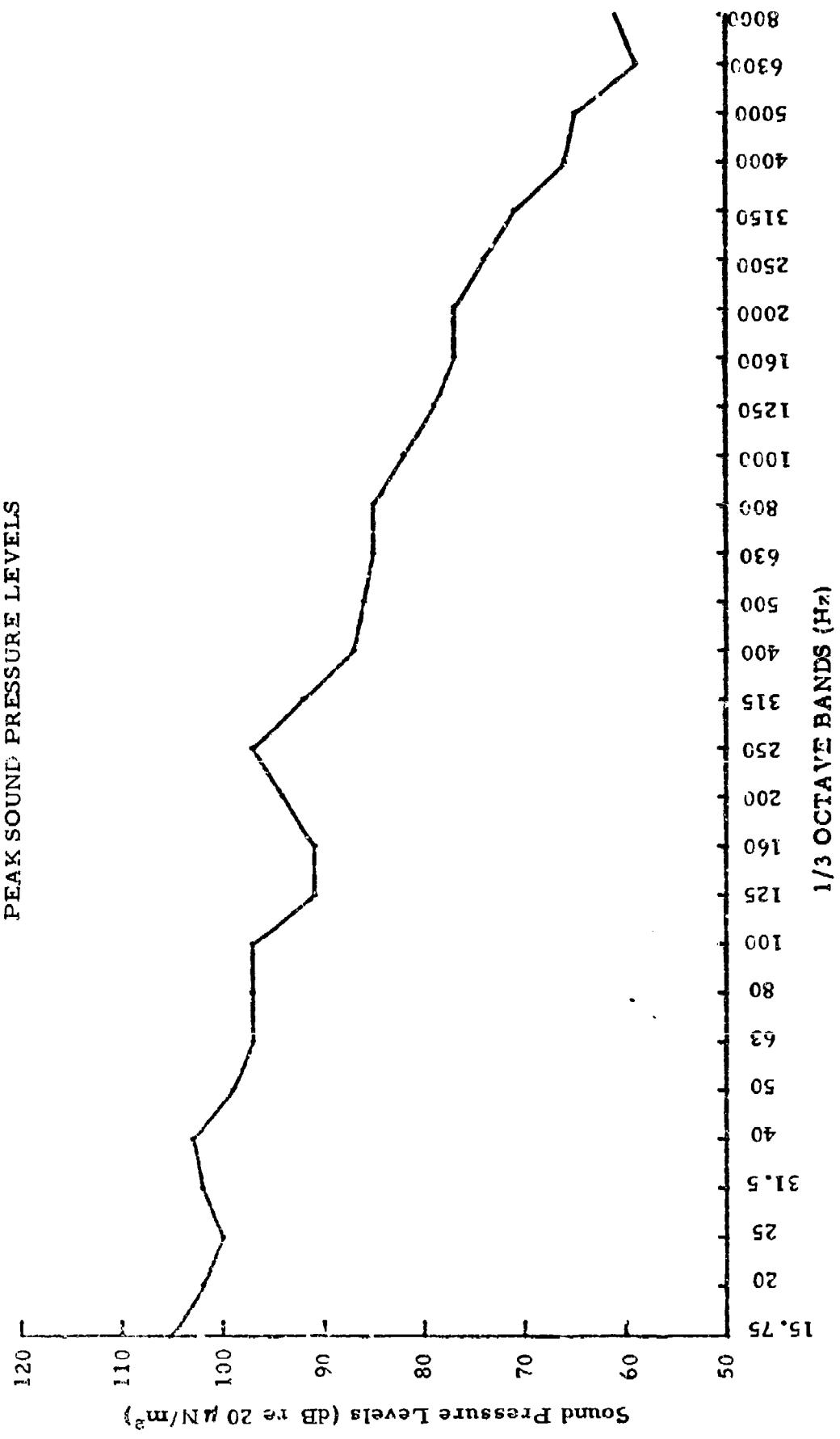


FIGURE 8  
OAK MOUNTAIN  
PEAK SOUND PRESSURE LEVELS

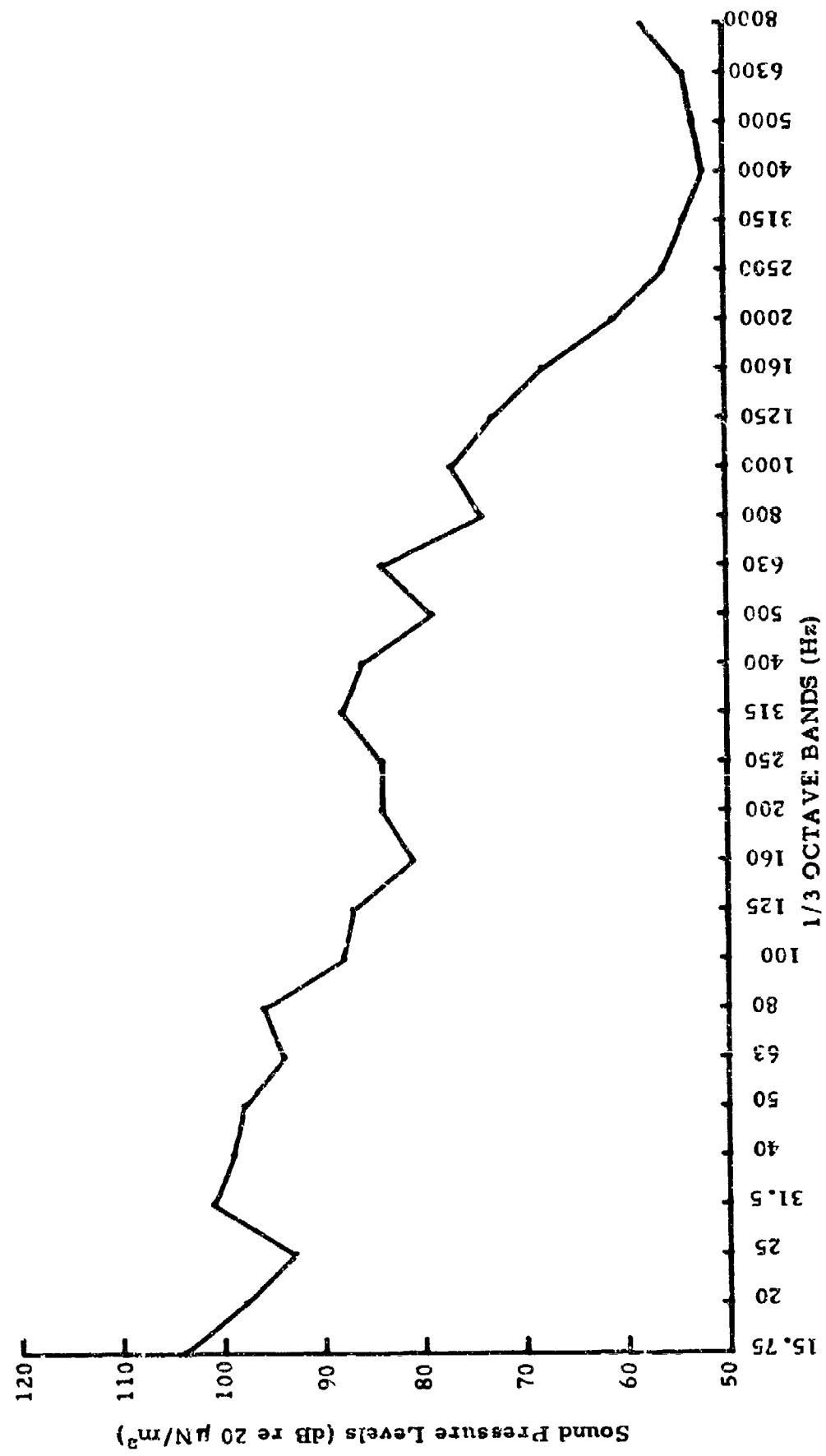
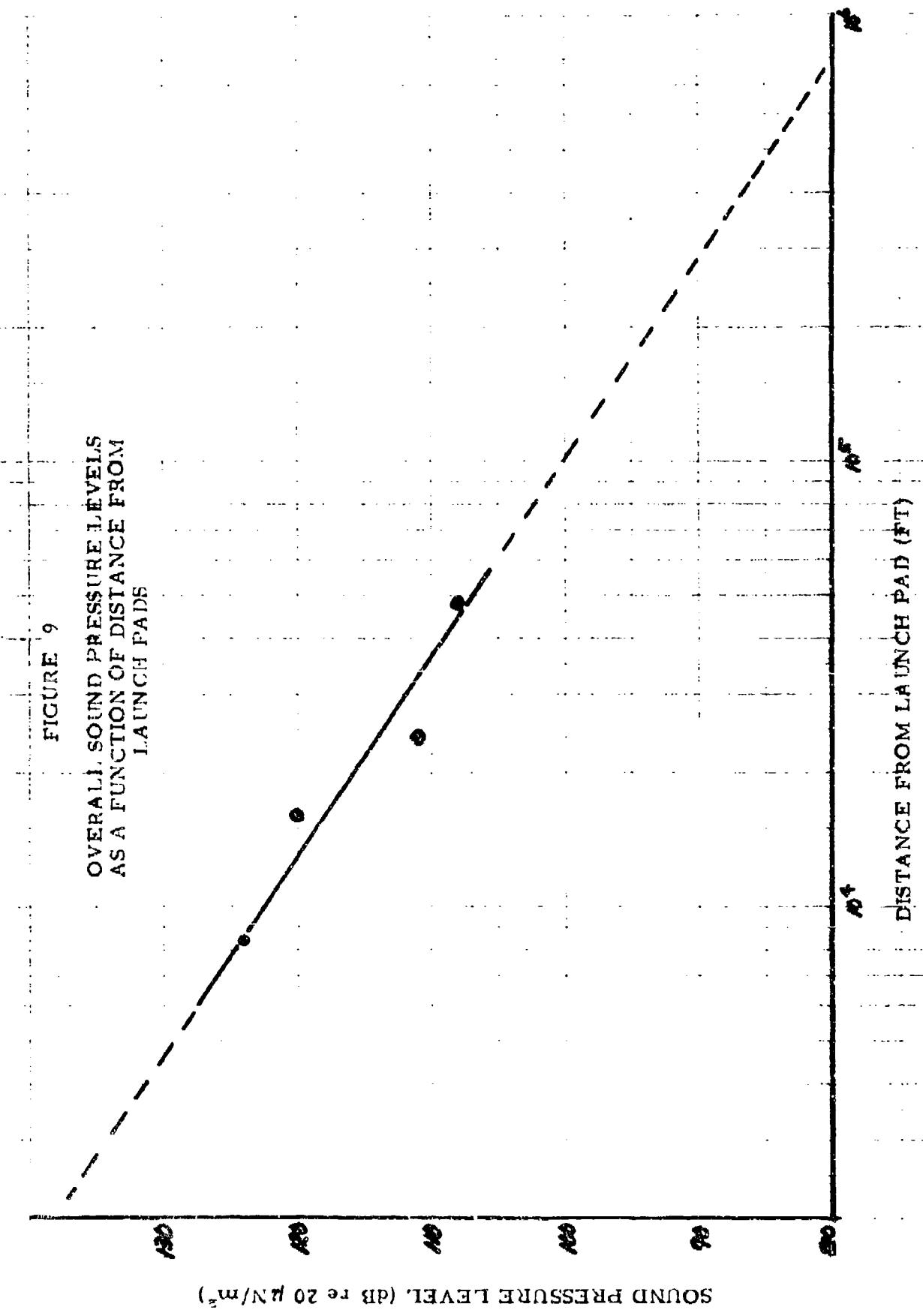


FIGURE 9

OVERALL SOUND PRESSURE LEVELS  
AS A FUNCTION OF DISTANCE FROM  
LAUNCH PADS



SOUND PRESSURE LEVEL (dB re 20  $\mu\text{N/m}^2$ )

sites and under different weather conditions should result in reasonably accurate predictions. However, the weather data during the measurements are shown in Appendix C so that correction for sound attenuation by air can be applied, if desired, when extrapolating these data to different weather conditions.

d. Estimated Environmental Impact:

(1) The impact of any single noise event is difficult to determine when one is concerned about levels and exposure times below those normally considered hazardous to hearing; however, perceived noise levels (PNL's) and effective perceived noise levels (EPNL's) are commonly used to define single event noise levels, e.g., aircraft flyovers. The maximum PNL's calculated from 1/3 octave band data (50 Hz - 10,000 Hz) are shown in conjunction with expected responses or judgements of personnel exposed in Table VII (Ref. 1).

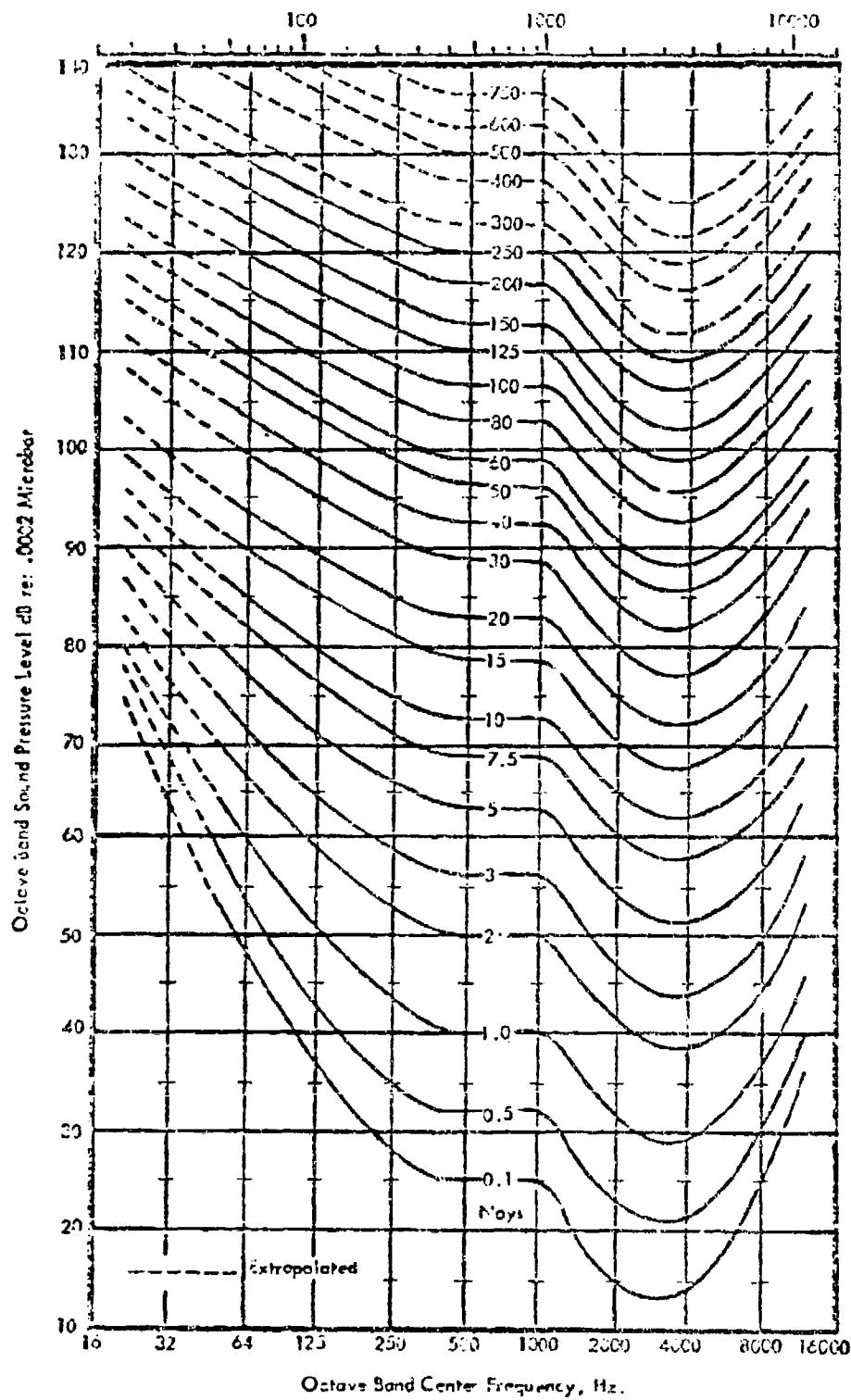
TABLE VII  
MAXIMUM PNL'S AND EXPECTED RESPONSE

<u>Location</u>	<u>Max. PNL (dB)</u>	<u>Ranges of Response</u>
SLC-3	118	Unacceptable, annoying, noisy
Range Operations	114	
Tranquillon Peak	110	Unacceptable, intrusive, moderately noisy
Oak Mountain	105	Barely acceptable, intrusive, moderately noisy

The PNL is calculated from a subjective unit of noisiness called the "noy." A sound of two noy is said to be subjectively twice as noisy as a sound of one noy. Curves of equal noisiness, with sound pressure levels plotted as a function of frequency, are shown in Figure 10. Considerably higher sound pressure levels are required in the lower frequencies than in middle and high frequencies to produce a subjectively equal level of noisiness. The PNL is calculated from these type curves and is basically a translation of the subjective noy scale to a dB scale. Therefore, the

FIGURE 10

Subjective Reaction to Acoustic Noise  
**CURVES OF EQUAL NOISINESS**



frequency or spectral characteristics of the noise significantly influence the value of the PNL calculated and may appear to be lower or higher than might be expected on the basis of overall sound pressure levels.

(2) The Noise Exposure Forecast (NEF) which is used to estimate community response to a series of noise events can be calculated from PNL's and EPNL's. The basic equations used for determination of EPNL and NEF are shown below:

$$EPNL = PNL + 10 \log \frac{t}{15}$$

where: PNL = maximum PNL calculated from 1/3 octave bands (50 Hz - 10,000 Hz) sound pressure levels

$t$  = duration, in seconds, of noise within 10 dB of maximum PNL

$$NEF = EPNL + 10 \log \left( \frac{Nd}{20} + \frac{Nn}{1.2} \right) - 75$$

where: Nd = number of day operations (0700 - 2200 hrs)

Nn = number of night operations (2200 - 0700 hrs)

The NEF at each site was determined by assuming one Titan III D missile launch per day. This is probably an overestimate of activity. These NEF's are shown for each measurement site in Table VIII.

TABLE VIII  
NEF'S AND ESTIMATED COMMUNITY RESPONSE

<u>Location</u>	<u>Approximate NEF's</u>	<u>Estimated Community Impact (Ref. 1)</u>
SLC-3	34	Individuals in private residences may complain vigorously.
Range Ops (Bldg 488)	30	Commercial use - OK
Tranquillon Peak	27	Satisfactory for all uses except possibly schools, churches, hospitals, etc.
Oak Mountain	23	

It should be realized that the NEF was developed primarily for aircraft noise after studying the reactions of large numbers of individuals to specific aircraft noises. Also, individual tolerance varies considerably as demonstrated by survey around London Heathrow Airport which determined that approximately 10 percent of the population were ultra-sensitive to noise and objected to any outside noise intrusion while 25 percent of the population were unaffected even by very high community noise levels (Ref. 1,2). Therefore, these two portions of the population (35%) will not be significantly affected by noise abatement efforts; only the remaining 65 percent of the population will be helped by noise control and characterizing the noise environment. Some of the important factors other than the actual noise environment which affect human response to noise intrusion include:

- (a) Socioeconomic status
- (b) Relative importance and necessity of noise source
- (c) Relation of noise source to individual, i.e., an individual who is economically dependent on the noise source is less likely to be annoyed
- (d) Activity being done during noise exposure

(3) Acoustical damage to structures can be estimated from Figures 11 and 12. No damage is expected at distances greater than 24,000 ft.

### 3. Vibration:

a. Peak Vibration and Frequency Distributions (La Purisima Mission): Only the peak vibration (acceleration) levels were reported as other levels were not significantly above background. The peak vibration levels occurred essentially simultaneously with the peak noise levels, 92 - 100 seconds after launch, indicating that the vibration was induced by airborne noise. Thus, noise at this site would also be a satisfactory measurement parameter for damage estimation as there are single event noise criteria relating noise levels at various frequencies to structure damage (see Figures 11 & 12). Peak and background acceleration from 25 Hz to 2000 Hz (limiting frequencies of measurement equipment) are shown in Tables IX and X.

#### b. Effects on La Purisima Mission:

(1) Examination of the acceleration data indicates that peak values were only slightly above background and in some frequencies

FIGURE 11

## ACOUSTICAL DAMAGE CRITERIA FOR WALLS

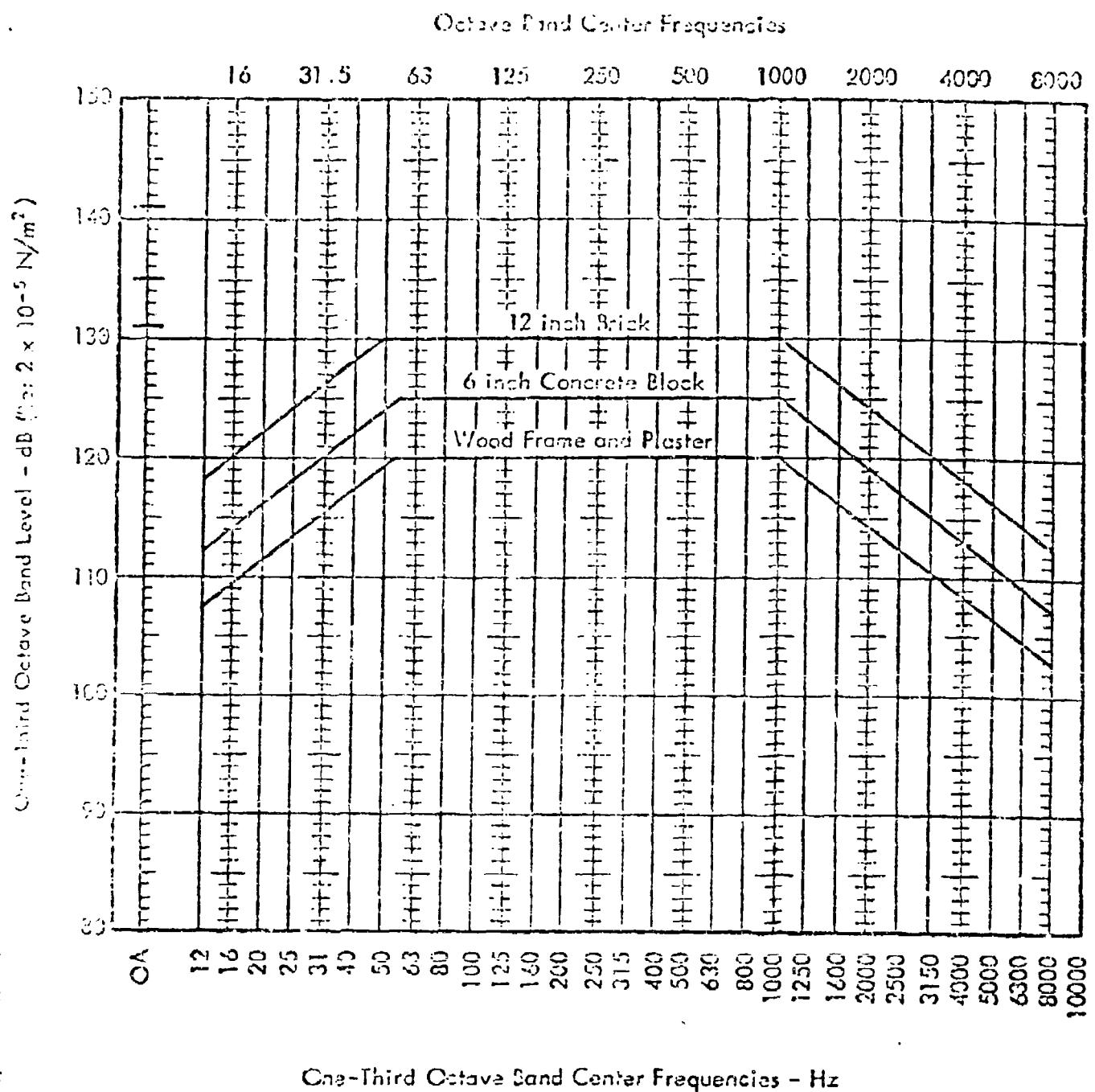
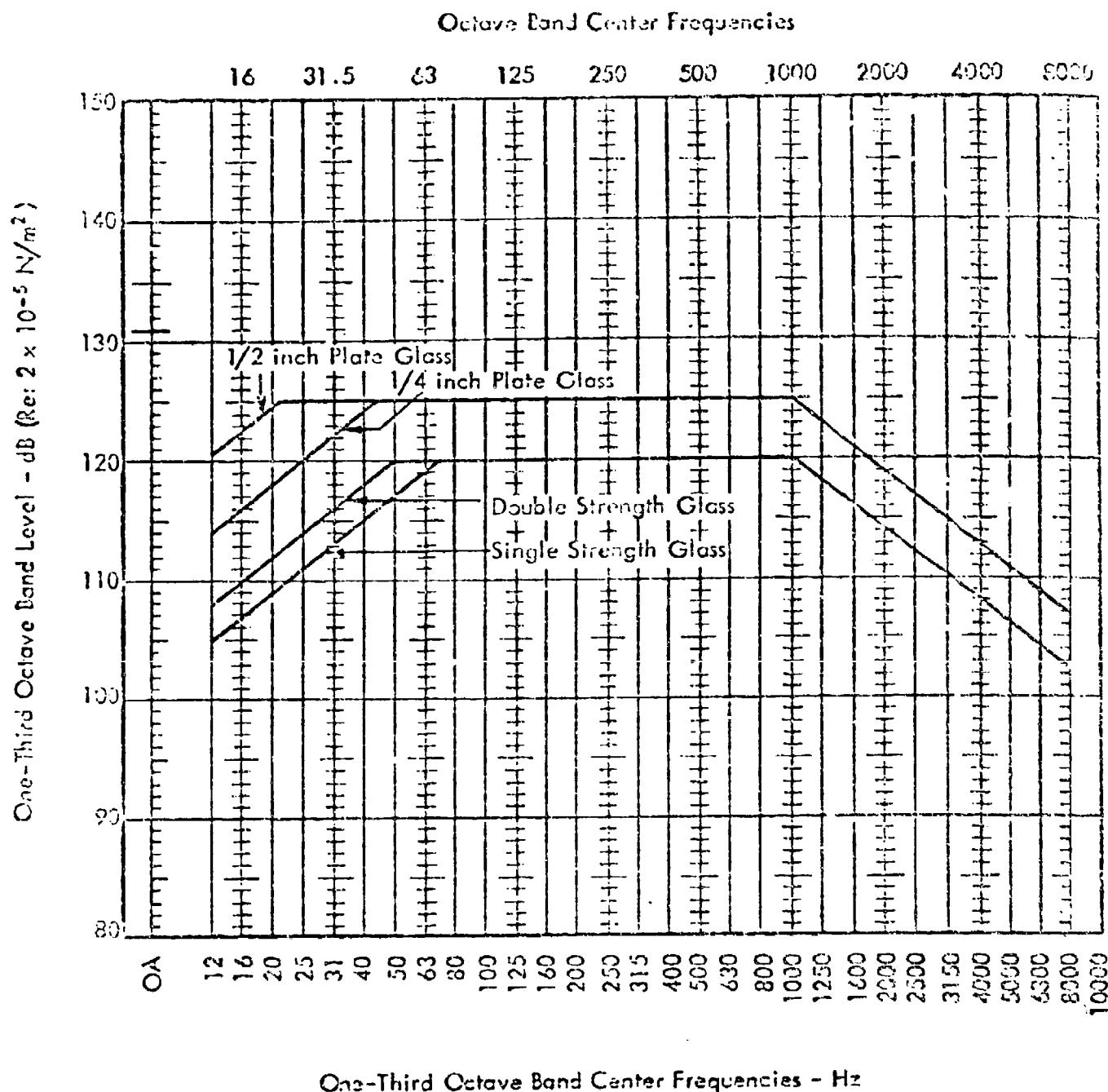


FIGURE 12

ACOUSTICAL DAMAGE CRITERIA FOR GLASS



**TABLE IX**  
**VIBRATION DATA**  
**ACCELEROMETER A**

<b>1/3 Octave Center Frequency (Hz)</b>	<b>Peak Acceleration (g's)</b>	<b>Background (g's)</b>
2.0K	$7.8 \times 10^{-5}$	$1.4 \times 10^{-4}$
1.6K	$9.8 \times 10^{-5}$	$2.8 \times 10^{-4}$
1.25K	$6.3 \times 10^{-5}$	$1.6 \times 10^{-4}$
1.0K	$5.7 \times 10^{-5}$	$7.5 \times 10^{-4}$
800	$4.9 \times 10^{-5}$	$5.0 \times 10^{-5}$
630	$4.5 \times 10^{-5}$	$5.8 \times 10^{-5}$
500	$5.8 \times 10^{-5}$	$8.2 \times 10^{-5}$
400	$1.5 \times 10^{-4}$	$1.7 \times 10^{-4}$
315	$1.5 \times 10^{-4}$	$1.9 \times 10^{-4}$
250	$1.8 \times 10^{-4}$	$8.0 \times 10^{-5}$
200	$3.8 \times 10^{-4}$	$1.3 \times 10^{-4}$
160	$1.6 \times 10^{-3}$	$4.2 \times 10^{-4}$
125	$4.1 \times 10^{-3}$	$4.9 \times 10^{-4}$
100	$6.7 \times 10^{-4}$	$1.6 \times 10^{-4}$
80	$5.3 \times 10^{-4}$	$6.7 \times 10^{-4}$
63	$2.2 \times 10^{-3}$	$4.8 \times 10^{-3}$
50	$1.1 \times 10^{-3}$	$1.2 \times 10^{-3}$
40	$2.0 \times 10^{-4}$	$4.2 \times 10^{-4}$
31.5	$2.1 \times 10^{-4}$	$2.8 \times 10^{-4}$
25	$4.0 \times 10^{-4}$	$1.4 \times 10^{-4}$

TABLE X  
VIBRATION DATA  
ACCELEROMETER B

<u>1/3 Octave Center Frequencies (Hz)</u>	<u>Peak Acceleration (g's)</u>	<u>Background (g's)</u>
2.0K	$2.9 \times 10^{-4}$	$1.1 \times 10^{-6}$
1.6K	$2.9 \times 10^{-4}$	$1.4 \times 10^{-6}$
1.25K	$2.9 \times 10^{-4}$	$1.1 \times 10^{-6}$
1.0K	$2.9 \times 10^{-4}$	$1.0 \times 10^{-6}$
800	$2.9 \times 10^{-4}$	$9.2 \times 10^{-6}$
630	$2.9 \times 10^{-4}$	$9.2 \times 10^{-6}$
500	$2.9 \times 10^{-4}$	$9.2 \times 10^{-6}$
400	$2.9 \times 10^{-4}$	$9.2 \times 10^{-6}$
315	$4.1 \times 10^{-4}$	$9.2 \times 10^{-6}$
250	$5.2 \times 10^{-4}$	$9.2 \times 10^{-6}$
200	$1.2 \times 10^{-3}$	$9.2 \times 10^{-6}$
160	$2.3 \times 10^{-3}$	$9.2 \times 10^{-6}$
125	$4.1 \times 10^{-3}$	$9.2 \times 10^{-6}$
100	$1.5 \times 10^{-3}$	$9.2 \times 10^{-6}$
80	$3.8 \times 10^{-4}$	$9.2 \times 10^{-6}$
63	$3.8 \times 10^{-4}$	$9.2 \times 10^{-6}$
50	$3.8 \times 10^{-4}$	$9.2 \times 10^{-6}$
40	$2.9 \times 10^{-4}$	$9.2 \times 10^{-6}$
31.5	$2.9 \times 10^{-4}$	$9.2 \times 10^{-6}$
25	$5.2 \times 10^{-4}$	$9.2 \times 10^{-6}$

the background fluctuated above peak levels.

(2) The magnitude of these acceleration values are roughly an order of magnitude or more below levels where individuals begin to perceive vibration. On this basis, the vibration would appear to be inconsequential. Seismic shock limits (Ref. 3) for building structures define a caution zone of approximately 0.01 g's which is well above the maximum acceleration level measured at the Mission. The measured vibration levels were also well below the structural damage threshold criteria adopted by the US Bureau of Mines (Ref. 4). If one assumes simple harmonic motion (sine wave) the displacement can be easily calculated from the acceleration values using the following relationship.

$$\text{Displacement} = \frac{\text{Acceleration} \text{ (inches/sec}^2\text{)}}{4 \pi^2 (\text{frequency})^2}$$

The resulting displacement values were orders of magnitude below displacement limits specified for safety from seismic shock damage.

(3) The Civilian Conservation Corps (CCC) restored the Mission ruins from 1934 to 1941. During the restoration certain construction modifications were made to make the buildings stronger, safer and more resistant to earthquakes. These modifications included the use of reinforced concrete columns, girders, beams and massive wooden beams. These construction modifications significantly reduce the vibration damage potential to the restored Mission as compared to the original adobe construction and the vibration levels measured during the missile launch would have little if any effect on the Mission.

#### SECTION IV

#### CONCLUSIONS

1. These noise data and similar data can be used to predict noise levels at distances other than those specified in this report and for future launches. The effects of varying atmospheric conditions are minimized because low frequency noise is not significantly affected by normal variations in atmospheric conditions; however, any extreme deviations from average atmospheric conditions should be considered when extrapolating these data.
2. No significant environmental impact is expected to result from these launches at distances of eight miles (44,000 ft) or greater from the

launch site. The relatively short duration of noise and infrequent occurrence of these launches further reduces the acoustical impact. These launches would probably be tolerated by personnel residing at closer distances, i.e., four miles from the launch pad.

3. The acceleration (vibration) levels, even-though unsophisticated and measured at only one location within the Mission, were so small that it is inconceivable that any damage to the La Purisima Mission buildings could result from the vibration caused by these launches. However, noise levels were also measured at the Mission by personnel of the Bioacoustics Division, Aeromedical Research Laboratories and these data should also be considered with regard to information presented in Figures 11 & 12 before any definite conclusions are drawn regarding the damage potential of these launches.

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**APPENDIX A**

**SIMPLIFIED UNCERTAINTY ANALYSIS  
OF  
RECORDING & ANALYSIS SYSTEMS**

A simplified uncertainty analysis was performed on the recording and analysis systems. Sources of error in recording include variations in microphone incidence angle, tape characteristics, non-linearity characteristics of the octave band analyzer, recorder, playback unit and real time analyzer, and calibration instrumentation uncertainties. The uncertainty in the incidence angle has the potential for the largest single error in the system as it seriously affects microphone sensitivity, especially at high frequencies. The limit of error calculation for the system is listed below. The final uncertainty (limit of error) in the data presented is the rms uncertainty of the components and was calculated to be  $\pm 1.9$  dB.

<u>PARAMETER</u>	<u>Limit of Error (dB)</u> <u>(RE: 20 <math>\mu</math>N/m<sup>2</sup>)</u>
Ampeax 434 uniformity (reel to reel)	$\pm 1.0$
Ampeax AG-350 reproducer (estimated non-linearity)	$\pm 0.1$
Uher 2000L recorder (estimated non-linearity)	$\pm 0.5$
1558 Filter uncertainty	1.0
Microphone Angle of Incidence	$\pm 1.0$
Real Time Analyzer non-linearity	$\pm 0.5$
<hr/>	
data limit of error = $\lambda$ =	$\pm 1.9$

**APPENDIX B**

**ACCUSTIC DATA TITAN III LAUNCHES  
(SAMSO/LVRO)**

DEPARTMENT OF THE AIR FORCE  
HEADQUARTERS SPACE AND MISSILE SYSTEMS ORGANIZATION  
LOS ANGELES AIR FORCE STATION, PO BOX 7100, WELDOWAY, LOS ANGELES, CALIFORNIA 90009

LIVRC

20 AUG 1974

Acoustic Measurements on Titan III Launches

WELL, JR

1. As part of a DOD-directed study on SCS siting at Vandenberg AFB, we are reviewing the limitation imposed on potential sites by acoustic energy levels.
2. To obtain current acoustic data on launch vehicle systems similar to the Space Shuttle, request you obtain data per the attached requirements document on the next Titan IIIC, IIID, and IIIE launches. We understand that these data can be obtained through a currently planned measurements program and, therefore, at no cost to the program office.
3. Three copies of the resulting data and analyses (overall and octave band sound pressure level histories vs time and background level) should be forwarded to SAMSO/LVRC, Maj A.B. Sloan, Autoven 663-1570.

*Kerry W. Haiker*

KERRY W. HAIKER, Lt Col, USAF  
Test Program Director, Ops & Eval  
Reusable Launch Vehicles SPO

1 Atch  
Acoustic Measurements  
Requirements

Cy to: AMRT/DO  
AMRT/DEM  
S. 100/DAM  
6555/AMC/OC  
6595/SLC/OC  
Aerospace/J. Smith  
Aerospace/P. Tornonova  
Aerospace/R. Kendall

# ACOUSTIC MONITORING AND RECORDING

## 1. PURPOSE

To outline requirements for monitoring and/or monitoring instrumentation for Titan IIIC, IIID, and IIE vehicles at range, C. Averell Hill and Vandenberg AFB.

## 2. SCOPE

This chapter covers the identification of acoustical and meteorological instruments and analytical methodology to obtain and evaluate overall short-term-acoustic sound pressure levels in addition to range time and space distributions for time periods of particular interest. Instantaneous slices are recommended to document sound propagation. Launch site distances from launch sites and for obtaining Lempco area environmental sound levels.

## 3. AUDIOPHYSICAL

### 3.1 Monitoring

In audio pick-up/recording, one is required for each observation station. Audio instrumentation need not be identical if each station can have similar operating characteristics and be calibrated. The absolute recorded data referenced to a common standard may be used compatible station-set to station-set. This is essential to eliminating pick-up/recording instrumentation as a source of discrepancy in observation. A positive correlation station-to-station and with range time must be established at rocket ignition and be maintained throughout vehicle lift-off.

Figure 1 represents the estimated overall sound pressure level expected to be generated by Titan IIIC, IID, and IIE vehicles launched at different distances from the launch site. The approximate spectral content of the launch sound is shown in Figure 2. Figures 1 and 2 are to be used in evaluating and selecting appropriate audio pick-up and recording components.

### 3.0 Acoustic Monitoring

For accurate and sensitive band sound pressure level/time histories, appropriate calibrations must be maintained to assure proper amplitude levels from the recorded data. It is important that amplitude and frequency correlate sound levels with real time.

#### 3.0.1 Measurement Setting - VAPR

Proposed acoustic measurement sites are shown in Figure 3 for the VAPR/Lempke area. Sites situated roughly on four radials from S.C.-1 have been indicated. Most of these have been located at existing installations to take advantage of existing electric power for the pick-up/recording sets (although battery-powered sets will also be provided, if needed) and to simplify coordination of recorded data with range time.

The available recording sets are such that up to five recording sites can be used simultaneously. Priorities have been assigned to the suggested sites as follows:

Priority	Site
1	Mr. Furkiss Mission
2	Radial 2
3	Radial 4
4	Radial 1
5	Radial 3

SAMP is publicly committed to making acoustic observations at Mr. Furkiss Mission. Following that, priority is given to recording data at a minimum of three different distances from the plant, and preferably along the same radial. Since the existence of background sound level data from the VAPR/Lempke area is unknown, suggested sites are indicated on Figure 3 to be instrumented singly or simultaneously over a period of several weeks to characterize prevailing neighborhood noise.

### 5.5 AEROMETRIC STATIONS - CLASS

Aerometric measurements will be used by NASA for monitoring plume dispersion and downwind plume characteristics. These data will be obtained in accordance with the following procedures:

- a. Downwind plume monitoring will be conducted at the following measurement sites and provide aerometric data....

### 6.0 INSTRUMENTATION REQUIREMENTS

At each VAB and VAFB, print-outs of wind systems data observed at selected stations during a vehicle launch period will be required. Specific data required at lift-off are:

- a. Wind Speed ....., each station
  - b. Wind Direction ....., each station
  - c. Temperature ....., each station
  - d. Barometric Pressure ....., as available
  - e. Relative Humidity ....., as available
  - f. Downwind flow data (vertical wind, temperature, density, helium concentration).
  - g. Solar radiation ....., as available
- In addition to the above, staff meteorological prints will be required at the more significant observation stations.

MAX OVERALL SOUND PRESSURE LEVEL - dB re:  $2 \times 10^{-5}$  N/m<sup>2</sup>

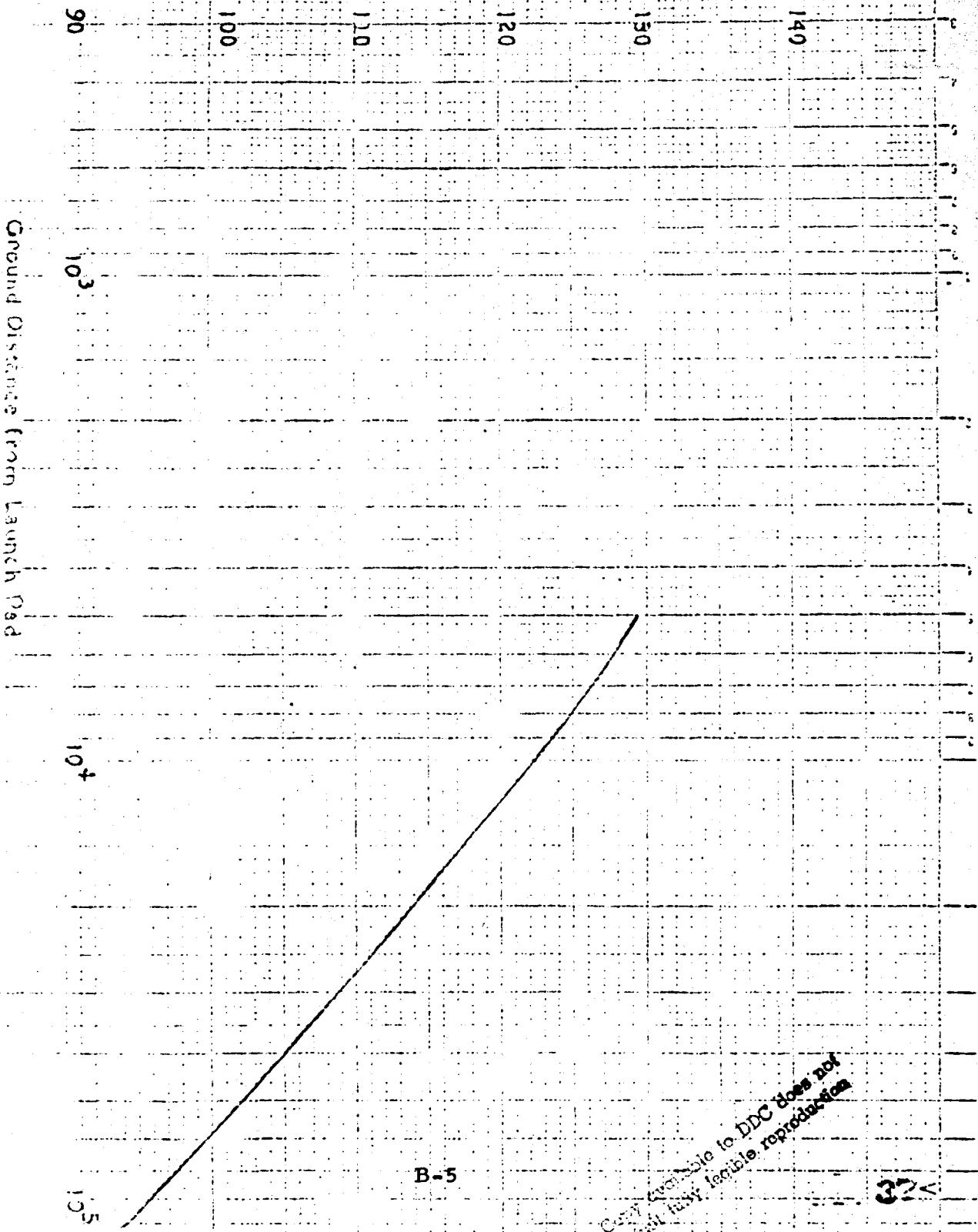
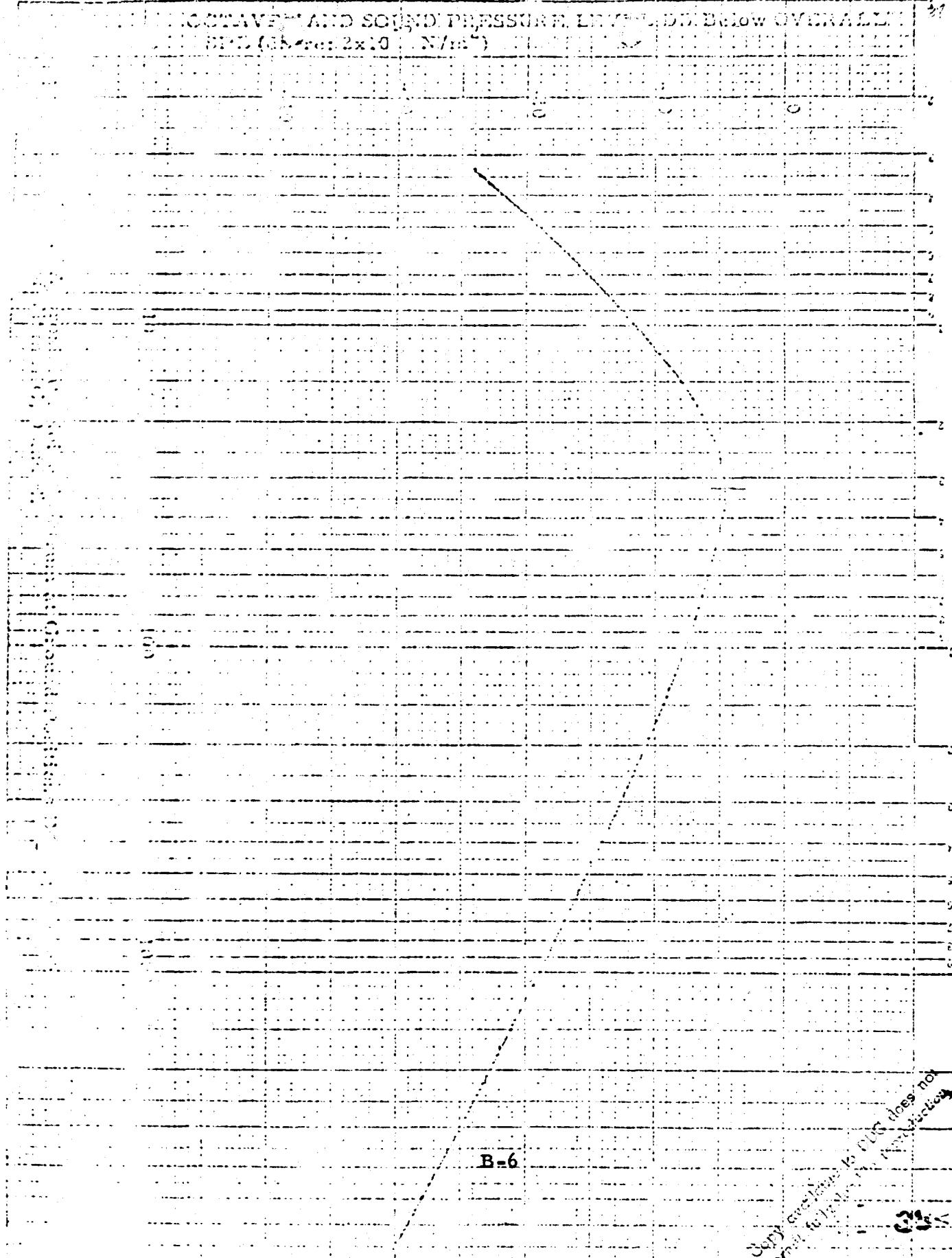


FIG. 1 Estimated T-33C, D-5 Launch Overall Sound Pressure Levels vs.  
Ground Distance from Launch Pad.

11 OCTAVE AND SOUND PRESSURE LEVELS DB BELOW OVERALL  
SPL (Ref. 2x10<sup>-5</sup> N/m<sup>2</sup>)



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necessarily reflect actual subject

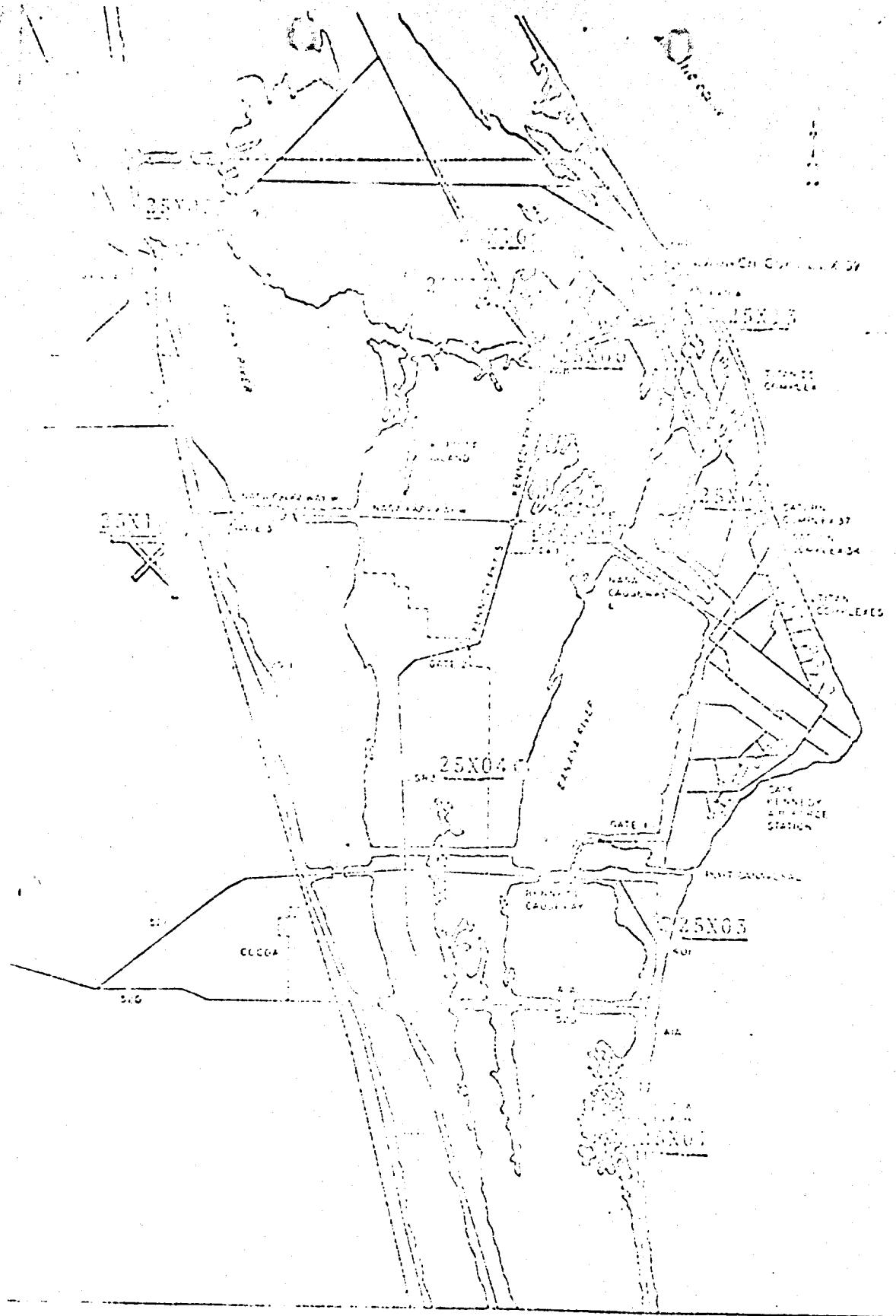


Figure 11. Far Field Acoustic Measurements Locations B-7 CCAPS

**APPENDIX C**

**WEATHER DATA 29 OCT 74  
VANDENBERG AFB**

MMWVZCZG000070

RTTURYUW VI

3799 J022236-UUUU-0000

ZNR UUUUU

ZNR UUUUU

R 292238Z OCT 74

JN VEA VAFB CA

TO AFSCF (DOW) SUNNYVALE CA

ASH BLDG 21130

BT

TEST NBR 07122 T-8  
RAVINSIDE RUN AN/ORD-1  
BLDG 900 - VAFB, CALIF.  
1948Z 29 OCT 1974  
ASCENT NBR 8189

WIND 10 KTS 1000 FT

ALT (FT)	DIR	KTS	TEMP	D/PT	PRESS	RH	AB/H	DEW	IR	V3	SHP	SVD
000367	300	013	14.3	6.9	8995.00	868	87.90	1280.72	313	661	8666	866
001500	319	017	12.1	8.7	8972.51	850	88.57	1182.39	316	658	.812	882
002600	328	016	9.2	6.8	8937.71	866	87.18	1158.48	301	635	.882	113
003600	329	017	6.6	2.5	8983.88	874	85.66	1121.26	285	632	.884	837
004600	337	016	4.8	1.3	8878.82	879	63.32	1688.38	276	649	.885	838
005600	348	023	2.7	.9	8838.84	856	82.11	1956.43	268	647	.811	349
006600	337	036	1.6	-2.4	8867.83	873	84.89	1821.96	254	646	.889	319
007600	331	033	-.3	-6.5	8777.66	869	82.98	8938.68	239	644	.888	293
008600	327	032	-1.8	-8.7	8748.84	836	82.54	8937.18	238	643	.884	283
009600	335	027	-1.2	-10.3	8738.83	847	82.13	8922.91	219	642	.811	114
010600	338	025	-2.7	-11.7	8693.88	858	82.80	8898.58	218	641	.883	829
011600	336	032	-4.8	-12.8	8667.61	834	81.83	8845.39	203	638	.888	325
012600	334	036	-5.7	-14.9	8642.83	845	81.96	8835.79	196	637	.887	315
013600	338	037	-7.6	-16.4	8617.72	849	81.39	8829.33	183	635	.882	277
014600	331	038	-9.9	-18.2	8593.96	831	81.28	8785.44	123	638	.882	275
015600	333	039	-12.3	-19.5	8578.98	837	81.88	8761.77	177	629	.884	813
016600	337	044	-14.9	-21.4	8548.79	837	80.32	8739.36	171	626	.888	818
017600	341	045	-17.2	-23.9	8526.88	835	80.73	8716.56	165	623	.889	839
018600	344	053	-19.3	-26.4	8503.84	834	80.68	8698.97	159	621	.889	828
019600	346	056	-23.8	-27.2	8485.47	823	80.96	8673.15	134	637	.883	866
020600	347	063	-24.7	-28.6	8469.73	821	80.35	8652.76	136	614	.812	355
021600	351	062	-27.1	-29.7	8446.61	879	80.46	8632.14	144	611	.813	836
022600	353	063	-29.3	-32.1	8428.85	877	80.37	8611.38	139	608	.811	152
023600	352	061	-32.3	-36.2	8410.13	864	80.23	8595.11	134	605	.803	195
024600	351	061	-34.4	-42.6	8395.75	844	80.13	8578.18	129	602	.808	203
025600	358	061	-38.8	-45.8	8373.59	834	80.06	8554.23	124	599	.808	256
026600	348	061	-39.3	99.9	8358.76	781	92.32	8536.32	120	796	.808	891
027600	348	062	-42.5	99.9	8344.85	779	99.99	8519.64	116	792	.803	868
027600	348	064	-43.4	99.9	8328.81	773	99.99	8502.91	112	588	.808	838
029600	348	069	-48.1	99.9	8314.89	799	99.99	8486.23	108	584	.808	826
030600	349	069	-51.1	99.9	8299.86	799	99.99	8476.42	103	581	.808	838
031600	349	065	-59.6	99.9	8286.17	789	99.99	8467.95	100	581	.801	158
032600	348	074	-51.4	99.9	8278.45	790	99.99	8459.96	896	588	.816	357
033600	347	083	-48.1	99.9	8268.76	799	92.33	8463.67	598	584	.814	337
034600	345	077	-43.6	99.9	8249.09	793	99.99	8451.48	585	588	.818	185
035600	343	071	-46.6	99.9	8237.98	799	99.99	8435.98	582	584	.813	189
036600	339	063	-48.9	99.9	8227.29	799	92.99	8393.18	579	583	.811	198
037600	337	063	-49.4	99.9	8217.61	799	99.99	8337.94	575	583	.806	218
038600	339	066	-49.3	99.9	8207.19	790	99.99	8322.92	572	583	.806	297

